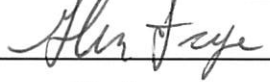


1. Approving Civil Aviation Authority Transport Canada		2. AUTHORIZED RELEASE CERTIFICATE FORM ONE			3. Form tracking Number: ACS Repair Order #: 9072 Revision #:	
4. Organizations Name and Address: ADVANCED COMPOSITE STRUCTURES, INC. 43 Muir Road, Winnipeg, MB., Canada R2X 2X7 Ph: (204) 982-6565 Fax: (204) 982-6555				5. Work order/Contract/Invoice Customer P.O. Number: R2020-62B		
6. Item: 1.	7. Description: - Tail Rotor Blade	8. Part Number: 212-010-750-105FM	9. Quantity: 1	10. Serial/Batch Number: A-11845	11. Status/Work: - Repair	
12. Remarks:						
BHT.212.TR.01.101		BHT-212-MM-7, Sec 64-45	Receiving Inspection		1. Inspected IAW ASB# 212-09-134 R-A / FAA AD# 2020-12-10. No defects found.	
BHT.212.TR.01.113		ASB #212-00-111 Rev.D	Installation or Replacement of Aft Tip Cap Rivets		2. Inspected IAW ASB# 212-13-147. No defects found.	
BHT.212.TR.01.104		ACS.SPM.GP.01.001 ASB# 212-05-122, BHT-212-CR&O, Section 64-00-00, Para 64-35	Replacement of Buffer/Fretting Pads			
BHT.212.TR.01.106		ACS.SPM.GP.01.001 BHT.212.TR.01.202 ASB 212-00-111 Rev.D	Tip Cover Replacement			
BHT.212.TR.01.107		BHT-212-CR&O, 64-00-00, Section 64-37 BHT.SPM.RB.01.001	Refinish Complete			
BHT.212.TR.01.115		ACS.SPM.RB.01.002	Static Balance			
TSN: 515.55						
13a. Certifies that the items identified above were manufactured in conformity to:				14a. <input checked="" type="checkbox"/> Car 571 Maintenance Release		
<input type="checkbox"/> Approved design data and are in condition for safe operation.				<input type="checkbox"/> Other registrations specified in block 13.		
<input type="checkbox"/> Non-approved design data specified in block 12.				Certifies that unless otherwise specified in block 12, the work identified in block 11 and described in block 12, has been performed in compliance with the Canadian Aviation Regulations		
13b. Signature		13c. Approval Organization Number		14b. Signature		14c. Approved Organization Number
						AMO # 39-89
13d. Name		13e. Date (dd/mmm/yyyy)		14d. Name		14e. Date (dd-mmm-yyyy)
				Glen Frye		29-Jun-2020
30 Dec 2008(Previously form 24-0078)						
Installer Responsibilities						
This certificate does not constitute authority to install. Installers working in accordance with the national regulations of a country other than the authority specified in Block 1, must ensure that their regulations recognize certifications from the country specified. Statements 13a or 14a do not constitute installation certification. In all cases, the aircraft technical record must contain an installation certification issued in accordance with the applicable national regulations before the aircraft is flown.						

NOMENCLATURE TAIL ROTOR BLADE
 PART NUMBER 212-010-750-105FM
 SERIAL NUMBER A-11845

NOTICE: WHEN RETURNING ANY LIMITED LIFE COMPONENT TO BHTC/BHTI FOR ANY REASON IT IS MANDATORY THAT A VALID HISTORICAL SERVICE RECORD ACCOMPANY THE COMPONENT

HISTORICAL SERVICE RECORD

INSTALLATION DATA						SCHEDULED REMOVAL		REMOVAL DATA						
DATE	INSTALLED ON A/C NO.	BY (ACTIVITY)	INSTALLED AT	COMPONENT HOURS/RIN				INSP. HRS	RETIR. HRS / YEARS	DATE	REMOVED AT	COMP HRS	COMP OH	REASON FOR REMOVAL
			A/C HOURS	SINCE NEW	SINCE O/H						A/C HOURS	SINCE NEW	SINCE O/H	
				HRS COMP. HRS	HRS COMP. HRS							HRS	HRS	
19-Jun-2001	36221		858.41	0.0	0.0	5000	/	23-Oct-2001	1304.47	446.06	N/A	Vibracion		
13-Sep-2003	36201	ASESA	2499.17	446.06	NEW	7053.11	/	11-Oct-2005	2569.06	515.55	N/A	Delaminacion en punt		
							/							
							/							
							/							
							/							
							/							

TECHNICAL DIRECTIVES AND HISTORY OF OVERHAUL

COMPONENT TOTAL TIME/RIN	DESCRIPTION OF WORK ACCOMPLISHED	SIGNATURE AFFIRMS WORK PERFORMED IN ACCORDANCE WITH APPLICABLE AIRWORTHINESS STANDARD		
		SIGNATURE	NO. CERT. NO	DATE
515.55	Inspected, Rebonded aft tip cap, Replaced tip cap rivets, Replaced fretting Pads, Refinish,	-----	-----	-----
-----	Balanced. All work for RO# 9072 carried out by ACS, AMO# 39-89	<i>Alan Lopez</i>	-----	29-Jun-2020
515.55	Inspected LAW SB# 212-09-134 + 212-13-147. No defects found.	<i>Alan Lopez</i>	-	29-Jun-2020

REFER TO APPLICABLE ASB, ADS AND SECTION 4 OF THE MAINTENANCE MANUAL FOR COMPONENT AIRWORTHINESS LIMITATIONS AND RIN(Retirement Index Number).

COMPONENT NAME BLADE, T/R
 PART NUMBER 212-010-750-105
 SERIAL NUMBER A-11845
 PAGE 1 OF 1

Bell Helicopter TEXTRON

A Subsidiary of Textron Inc.
 POST OFFICE BOX 482 • FORT WORTH, TEXAS 76101

NOTE: When returning any limited life component to (RNTI) for any reason, it is mandatory that a valid HISTORICAL SERVICE RECORD accompany component.


HISTORICAL SERVICE RECORD

INSTALLATION DATA							REPAIR SCHEDULED	REMOVAL DATA				
DATE	INSTALLED ON AIRCRAFT SERIAL NO.	BY (ACTIVITY)	INSTALLED AT A/C HRS. RIN	COMP. HRS./RIN		COMP. HRS. RIN	<input type="checkbox"/> INSP <input type="checkbox"/> O/H <input checked="" type="checkbox"/> RET	DATE	REMOVED AT A/C HRS. RIN	COMP. HRS./RIN		REASON FOR REMOVAL
				SINCE NEW	SINCE O/H					SINCE NEW	SINCE O/H	
				COMP. HRS.	COMP. HRS.					COMP. HRS.	COMP. HRS.	
19-Jul-01	1A-FFD 30021	3	858.91	0.0	0.0	5,000 HRS.		23/Oct/01	1304:47	446:06	NIT	Vibración de alta frecuencia.
13-Sep-03	36201	Asesa DGAC#53	2499.17	446.86	NEW	7053.11		21-X-2005	2,569.06 N/A	515:55	TSN	DELAMINACION EN PUNTA

HISTORY OF INSPECTION, OVERHAUL, REPAIR AND APPLICATION OF TECHNICAL BULLETIN, SERVICE BULLETIN, AIRWORTHINESS DIRECTIVES, ETC.

COMPONENT TOTAL TIME/RIN	DESCRIPTION OF WORK ACCOMPLISHED	COMPLIED WITH BY		
		SIGNATURE	CERT. NO.	DATE
T.T. 0:0 05.00-083	Se verificó A.S.B. 412-00-106, No aplica por el Numero de Serie. T.T. 0:0		DGAC #53	26-MAR-01
446.06	Inspected and found to be repairable, install new tip closure, install buffer pads X4 IAW 204-099-223 Vol. X Rev L, refinish per BPS 4488, Balanced to master P/N 212-010-750-009 S/N A-5367. All repairs done IAW Bell and RBI Specs		R8DR 7131	4-11-02

Refer to applicable regulatory agency direction and Section 4 of the Maintenance Manual for RIN Accumulation Instructions (Model 204 refer to Section 1).

1. Approving National Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG					3. Form Tracking Number: 11845	
4. Organization Name and Address: Rotor Blades, Inc. 113 Ida Road Broussard, Louisiana 70518						5. Work Order/Contract/Invoice Number: 11845		
6. Item:	7. Description:	8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:		
1	Tail Rotor Blade	212-010-750-105	212	1	A-11845	Repair		
13. Remarks: Inspected and found to be repairable, install new tip closure, install buffer pads X4 IAW 204-099-223 Vol. X Rev L, re:finish per BPS 4488, Balanced to master P/N 212-010-750-009 S/N A-5367. All repairs done IAW Bell and RBI Specs T/T 446.06								
14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.				19. <input type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.				
15. Authorized Signature:		16. Approval/Authorization No.:		20. Authorized Signature: <i>Carlos Salas</i> 		21. Approval/Certificate No.: R8DR713Y		
17. Name (Typed or Printed):		18. Date (m/d/y):		22. Name (Typed or Printed): Carlos Salas		23. Date (m/d/y): 04/11/02		
User/Installer Responsibilities								
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>								