

NOMENCLATURE MAIN ROTOR BLADE
 PART NUMBER 212-015-501-115
 SERIAL NUMBER A-2825

NOTICE: WHEN RETURNING ANY LIMITED LIFE COMPONENT TO BHTC/BHTI FOR ANY REASON IT IS MANDATORY THAT A VALID HISTORICAL SERVICE RECORD ACCOMPANY THE COMPONENT

HISTORICAL SERVICE RECORD

INSTALLATION DATA						REMOVAL DATA											
DATE	INSTALLED ON A/C NO.	BY (ACTIVITY)	INSTALLED AT	COMPONENT HOURS/RIN			SCHEDULED REMOVAL			DATE	REMOVED AT	COMP HRS		COMP OH		REASON FOR REMOVAL	
				SINCE NEW	HRS COMP. HRS	SINCE O/H	INSP. HRS	RETIR. HRS / YEARS	A/C HOURS			SINCE NEW	SINCE O/H				
														HRS COMP. HRS	O/H HRS		RETIR. HRS / DATE
7-Apr-2009	N213SJ	JDF	4208.6		293.3			/		23-Feb-2001	4544.9	557.6					
23-Feb-2001	N212SJ		5678.0		557.6			/		11-Apr-2009	8595.0	3474.6					
								/									
								/									
								/									
								/									
								/									

TECHNICAL DIRECTIVES AND HISTORY OF OVERHAUL

COMPONENT TOTAL TIME/RIN	DESCRIPTION OF WORK ACCOMPLISHED HISTORY OF INSPECTION, OVERHAUL, REPAIR AND APPLICATION OF TECHNICAL BULLETIN, SERVICE BULLETIN, AIRWORTHINESS DIRECTIVES, ETC.	SIGNATURE AFFIRMS WORK PERFORMED IN ACCORDANCE WITH APPLICABLE AIRWORTHINESS STANDARD		
		SIGNATURE	NO. CERT. NO	DATE
3474.6	Inspected Blade IAW BHT-212-CR&O-3, 62-9, repaired 1" smooth dent at Sta. 89 and skin corrosion measuring 0.2910 inch Dia at Sta. 127, both on bottom of blade, work done IAW BHT-212-CR&O-3 62-14 steps 1-11. Complied with AD 2011-23-02 and ASB 212-08-030.			
	Refinished bottom blade surface IAW BHT-212-MM. Rotor Concepts LLC. WO # W10456	<i>Stephen J. J. J.</i>	5RCR618C	20-May-2020

REFER TO APPLICABLE ASB, ADs AND SECTION 4 OF THE MAINTENANCE MANUAL FOR COMPONENT AIRWORTHINESS LIMITATIONS AND RIN/Retirement Index Number.

1. Approving Civil Aviation Authority/Country: FAA/United States		2.		3. Form Tracking Number: 20052020	
4. Organization Name and Address: ROTOR CONCEPTS LLC. 727 N. Woodland Road Building H., Lakeside, AZ. 85929				5. Work Order/Contract/Invoice Number: W10456	
6. Item:		7. Description:		8. Part Number:	
1		MAIN ROTOR BLADE		212-015-501-115	
9. Quantity:		10. Serial Number:		11. Status/Work:	
1		A-2825		REPAIRED	
12. Remarks: Inspected IAW BHT-212-CR&O-3, 62-9, repaired 1" smooth dent at Sta. 89 and skin corrosion measuring 0.2910 inch Dia. at Sta. 127, both on bottom side of blade, work done IAW BHT-212-CR&O-3, 62-14 steps 1-11. Complied with AD 2011-23-02 and ASB 212-08-03. Refinished bottom side of blade surface TSN: 3474.6. TSO: 3474.6. END.					
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.					
13b. Authorized Signature:		13c. Approval/Authorization No.:		14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.	
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14b. Authorized Signature: Stephen Farina	
				14c. Approval/Certificate No.: 5RCR618C	
				14d. Name (Typed or Printed): Stephen Farina	
				14e. Date (dd/mm/yyyy): 02/MAY/2020	
User/Installer Responsibilities It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					