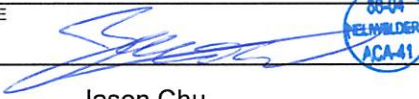



1. APPROVING CIVIL AVIATION AUTHORITY / COUNTRY Transport Canada		AUTHORIZED RELEASE CERTIFICATE FORM ONE			3. FORM TRACKING NO. 24-W7858-1	
4. ORGANIZATION NAME AND ADDRESS Heli-Welders Canada Ltd. Building #47, Unit 1, 21330 – 56th Ave, Langley Airport Langley, B.C. Canada V2Y 0E5				5. WORK ORDER / CONTRACT / INVOICE R2025-02D		
6. ITEM 1	7. DESCRIPTION Tail Rotor Assembly	8. PART NO. C642A0101052	9. QUANTITY 1	10. SERIAL / BATCH NO. M231	11. STATUS / WORK INSPECTED/TESTED	
12. REMARKS TC AD search carried out. EASA AD 2024-0209 (AH H120 ASB 64-21-0001 Issue 001) - Line of red paint added on each bolt and each nut of the link between tail hub and splined flange. 15FH recurring inspection due @ TSN: 4681.62 Hrs & 1000FH recurring inspection due @ TSN: 5666.62 Hrs. Previous certification: Form Tracking No. 24-W7631-1 Additional details of work accomplished are contained in Heli-Welders Canada Ltd W/O 24-W7858. TSN: 4666.62 Hrs						
13a. CERTIFIES THAT THE ITEMS IDENTIFIED ABOVE WERE MANUFACTURED IN CONFORMITY TO: <input type="checkbox"/> APPROVED DESIGN DATA AND ARE IN CONDITION FOR SAFE OPERATION <input type="checkbox"/> NON-APPROVED DESIGN DATA SPECIFIED IN BLOCK 12			14a. <input checked="" type="checkbox"/> CAR 571.10 MAINTENANCE RELEASE <input type="checkbox"/> OTHER REGULATIONS SPECIFIED IN BLOCK 12 <small>CERTIFIES THAT, UNLESS OTHERWISE SPECIFIED IN BLOCK 12, THE WORK IDENTIFIED IN BLOCK 11 AND DESCRIBED IN BLOCK 12 HAS BEEN PERFORMED IN COMPLIANCE WITH THE CANADIAN AVIATION REGULATIONS.</small>			
13b. SIGNATURE		13c. APPROVED ORGANIZATION NUMBER		14b. SIGNATURE 		14c. APPROVED ORGANIZATION NUMBER AMO 86-04
13d. NAME		13e. DATE (DD / MMM / YYYY)		14d. NAME Jason Chu		14e. DATE (DD / MMM / YYYY) 26/NOV/2024

Installer Responsibilities

This certificate does not constitute authority to install. Installers working in accordance with the national regulations of a country other than that specified in block 1 must ensure that their regulations recognize certifications from the country specified.

Statements in blocks 13a or 14a do not constitute installation certification. In all cases, the technical record for the aircraft must contain an installation certification issued in accordance with the applicable national regulations before the aircraft may be flown.

1. APPROVING CIVIL AVIATION AUTHORITY / COUNTRY Transport Canada		AUTHORIZED RELEASE CERTIFICATE FORM ONE			3. FORM TRACKING NO. 24-W7631-1	
4. ORGANIZATION NAME AND ADDRESS Heli-Welders Canada Ltd. Building #47, Unit 1, 21330 – 56th Ave, Langley Airport Langley, B.C. Canada V2Y 0E5				5. WORK ORDER / CONTRACT / INVOICE R2024-105D		
6. ITEM 1	7. DESCRIPTION Tail Rotor Assembly	8. PART NO. C642A0101052	9. QUANTITY 1	10. SERIAL / BATCH NO. M231	11. STATUS / WORK REPAIRED	
12. REMARKS <p style="text-align: center;">(EASA APPROVAL CERTIFICATE NUMBER: EASA.145.7157)</p> <p>1000HR/ 24M inspection carried out IAW Airbus ORION H120 Rev.009.00 MSM 05-25-00. Parts replaced: Tension Torsion Star P/N C642A0402101 S/N M932 TSN: 0.0 Hrs; Tail Rotor Blades Qty: 4 P/N C642A0300103 S/N M2113 TSN: 3458.3 Hrs, M3942 TSN: 3488.6 Hrs, M3949 TSN: 3894.4 Hrs, M3994 TSN: 3894.4 Hrs; Bushing Qty: 4 P/N 350A33-3010-20; Pin Qty: 8 P/N 7051A3642035 installed.</p> <p>TC/EASA AD search carried out. EASA AD 2021-0079 (AH H120 ASB 64A003 Rev 0) carried out satisfactory, no further action required; EASA AD 2021-0069 (AH H120 EASB 05A020 Rev 0) carried out satisfactory, hardware replaced. 15FH recurring inspection due @ 4681.64Hrs. 1000FH recurring inspection due @ 5666.62 Hrs. EASA AD 2018-0183 (AH H120 ASB 04A008 Rev 2) N/A per blade S/N's installed.</p> <p>Additional details of work accomplished are contained in Heli-Welders Canada Ltd W/O 24-W7631.</p> <p>Previous certification: EC120B, S/N 1281, VH-VTG TSN: 4666.62 Hrs</p>						
13a. CERTIFIES THAT THE ITEMS IDENTIFIED ABOVE WERE MANUFACTURED IN CONFORMITY TO: <input type="checkbox"/> APPROVED DESIGN DATA AND ARE IN CONDITION FOR SAFE OPERATION <input type="checkbox"/> NON-APPROVED DESIGN DATA SPECIFIED IN BLOCK 12			14a. <input checked="" type="checkbox"/> CAR 571.10 MAINTENANCE RELEASE <input checked="" type="checkbox"/> OTHER REGULATIONS SPECIFIED IN BLOCK 12 <small>CERTIFIES THAT, UNLESS OTHERWISE SPECIFIED IN BLOCK 12, THE WORK IDENTIFIED IN BLOCK 11 AND DESCRIBED IN BLOCK 12 HAS BEEN PERFORMED IN COMPLIANCE WITH THE CANADIAN AVIATION REGULATIONS.</small>			
13b. SIGNATURE		13c. APPROVED ORGANIZATION NUMBER	14b. SIGNATURE 		14c. APPROVED ORGANIZATION NUMBER AMO 86-04	
13d. NAME		13e. DATE (DD / MMM / YYYY)	14d. NAME Jason Chu		14e. DATE (DD / MMM / YYYY) 16/AUG/2024	

Installer Responsibilities

This certificate does not constitute authority to install. Installers working in accordance with the national regulations of a country other than that specified in block 1 must ensure that their regulations recognize certifications from the country specified.

Statements in blocks 13a or 14a do not constitute installation certification. In all cases, the technical record for the aircraft must contain an installation certification issued in accordance with the applicable national regulations before the aircraft may be flown.

INVENTAIRE DES COMPOSANTS
INVENTORY OF COMPONENTS

- 1- Date de pose - Date of installation
- 2- Numero de serie - Serial number
- 3- Heures de fonctionnement du composants
Operating time of component
- 4- Heures de fonctionnement de l'ensemble
Operating time of assembly
- 5- NB cycles du composant
Component number of cycles
- 6- NB cycles ensemble
Assembly number of cycles
- 7- Version - Version

POSE INSTALLATION (*)			
1	3	5	7
2	4	6	

DESIGNATION ENSEMBLE ASSEMBLY DESIGNATION	REFERENCE ENSEMBLE ASSEMBLY REFERENCE	NO DE SERIE SERIAL NUMBER	VERSION VERSION
MRA	C642 A0101052	M 231	

COMPOSANTS COMPONENTS

REFERENCE DESIGNATION	POSE INSTALLATION (*)			ECHANGES REPLACEMENTS (*)				Renseignements particuliers Special information
7050A3642044 AXE ETAGE DE FAISCEA SHOULDERED BOLT	2000/10	00000	/////					
	1307	00000	/////					
7050A3642044 AXE ETAGE DE FAISCEA SHOULDERED BOLT	2000/10	00000	/////					
	1308	00000	/////					
C642A0402102 Tension Torsion Star	2024-AUG	0.00	/					
	M932	4666.62	/					
C642A0300103 Equiped Blade	2006-MAY	0.00	/					
	M1796	2713.1	/					
C642A0300103 Equiped Blade	2006-MAY	0.00	/					
	M2080	2713.1	/					
C642A0300103 Equiped Blade	2006-MAY	0.00	/					
	M2082	2713.1	/					
C642A0300103 Equiped Blade	2006-MAY	0.00	/					
	M2103	2713.1 2713.1	/					
C642A0300103 Equiped Blade	16 AUG 2024	3458.3	/					
	M2113	4666.62	/					